

US006345792B2

(12) United States Patent Bilanin et al.

(10) Patent No.: US 6,345,792 B2

(45) **Date of Patent:** Feb. 12, 2002

(54) ACTUATING DEVICE WITH AT LEAST THREE STABLE POSITIONS

(75) Inventors: Alan J. Bilanin, Princeton; Robert M.

McKillip, Jr., Hopewell, both of NJ

(US)

(73) Assignee: Continuum Dynamics, Inc., Ewing, NJ

(US)

(*) Notice: Subject to any disclaimer, the term of this

patent is extended or adjusted under 35

U.S.C. 154(b) by 29 days.

(21) Appl. No.: 09/803,300

(22) Filed: Mar. 9, 2001

Related U.S. Application Data

- (62) Division of application No. 09/282,709, filed on Mar. 31, 1999, now Pat. No. 6,220,550.
- (60) Provisional application No. 60/080,154, filed on Mar. 31,
- (51) Int. Cl.⁷ B64C 13/24
- (52) **U.S. Cl.** **244/215**; 244/75 R; 244/213; 148/402; 114/332; 114/144 R

(56) References Cited

U.S. PATENT DOCUMENTS

2,624,531 A	1/1953	Stalker 244/17.13
2,642,143 A	6/1953	Miller 170/160.25
2,716,460 A	8/1955	Young 170/160.1
3,042,371 A	7/1962	Fanti 253/78
3,930,626 A	1/1976	Croswell, Jr 244/44

(List continued on next page.)

OTHER PUBLICATIONS

Chopra, I., "Development of a Smart Rotor," 19th European Rotorcraft Forum, Associazione Industrie Aerospaziali, Cernobbio (Como), Italy, Sep. 14, 1993, pp. N6–0 to N6–18. Duffy, R.E., et al., "A Theoretical and Experimental Study of the Snap-Through Airfoil and its Potential as a Higher Harmonic Control Device" (AA–88–0668), AIAA 26th Aerospace Sciences Meeting, Reno, NV, Jan. 11–14, 1988, pp. 1–11.

Giurgiutiu, V., et al., "Incrementally Adjustable Rotor–Blade Tracking Tab Using SMA Composites" (AIAA–97–1387), American Inst. of Aeronautics ans Astronautics, 1997, pp. 1456–1466

Hodgson, D., "Using Shape Memory for Proportional Control," *Engineering Aspects of Shape Memory Alloys*, Butterworth-Heinemann (1990), pp. 362–366.

Straub, F.K., et al., "Application of Smart Materials to Helicopter Rotor Active Control," SPIE vol. 3044, Symposium on Smart Structures & Materials, San Diego, CA, Mar. 3–6, 1997, pp. 99–113.

Yang, Y-J, et al., "Dynamics of a Bistable Snapping Microactuator," SPIE vol. 2443, Symposium on Smart Structures & Materials, San Diego, CA, 1995, pp. 754–762.

Primary Examiner—J. Woodrow Eldred

(74) Attorney, Agent, or Firm-David M. Quinlan, P.C.

57) ABSTRACT

A mechanical actuating device for moving an aerodynamic or hydrodynamic surface includes plural flexure members confined in an elastically deformed condition. The flexure members are movable against the force generated by their elastic deformation to move the device into one of at least three stable positions, in which the device, and therefore the aerodynamic or hydrodynamic surface, are held by the force generated by elastic deformation of the flexure members. Since the flexure members are always elastically deformed, the device "snaps" between discrete, stable positions and is held firmly in each. In another embodiment more flexure members can be used to provide additional stable positions. In one application, the actuating device is used as a trailing edge tab for a helicopter or tiltrotor blade to reduce 1/rev vibrations. The device can be actuated manually or electrically using shape-memory alloy wires to snap the flexure members into their various stable positions.

14 Claims, 4 Drawing Sheets

